

Apoqee Droque deploument

T+3.9 s

Time at motor burnout

91.2 ft/s

Velocity off rail

T-0s

Liftoff

# Hyperion

**Badger Ballistics** 

# **Team 143**

University of Wisconsin - Madison

#### Custom 3D Printed Avionics Bay & Booster Section COTS Fiberglass 5:1 VK Nosecone Main Parachute 10 ft CD 2.2 Ae rotech M2500 Air Brakes

1.100 ft

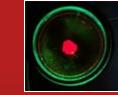
Main deployment

### **Vehicle Information**

Ve hicle Length [in]	Predicted Apoge e [ft]	Total Impulse [Ns]
144	11,154	9,671
Max Acceleration [G]	Motor	Max Velocity [ft/s]
9.67	M2500-T	1003

## **Payload**

A continuation from last year's experiment, the scientific payload attempts to measure a changing magnetic field using quantum photonics in the environment of a sounding rocket.



sustem acts as magnetometer with the following an engineered components: diamond with a nitrogen-vacancu center, a green light-emitting diode, a photodiode sensitive specifically to wavelengths of red light, and a Helmholtz coil.





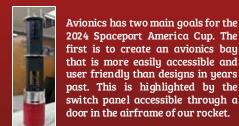
When hit with light from the LED, the nitrogen deficient diamond disperses a red light with an intensity proportional to the strength of the surrounding magnetic field. By measuring the intensity of this red light, the magnetic field strength can be extracted and compared to the true value of the magnetic field

The goal is to determine whether these measurements correlations can be made in a sounding rocket environment.



#### **Avionics**

T+250s





The second goal is to develop and implement reactive airbrakes to give altitude control for our rocket. Hyperion uses an SRAD flight computer to predict its apogee during flight and adjust the air brakes accordingly to reach the target apogee of 10.000 ft.

